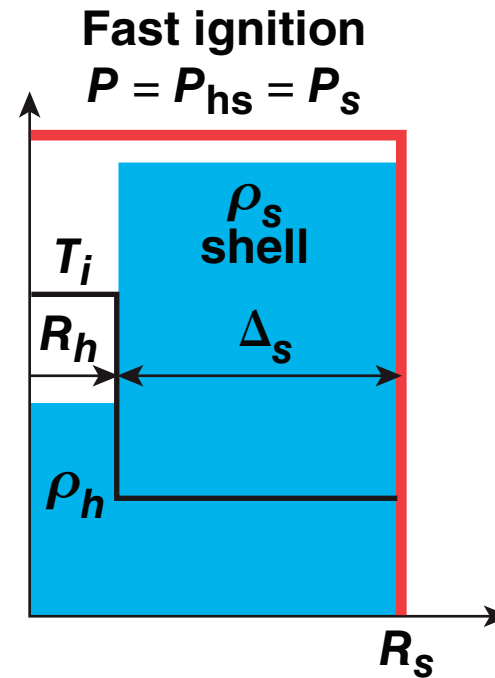
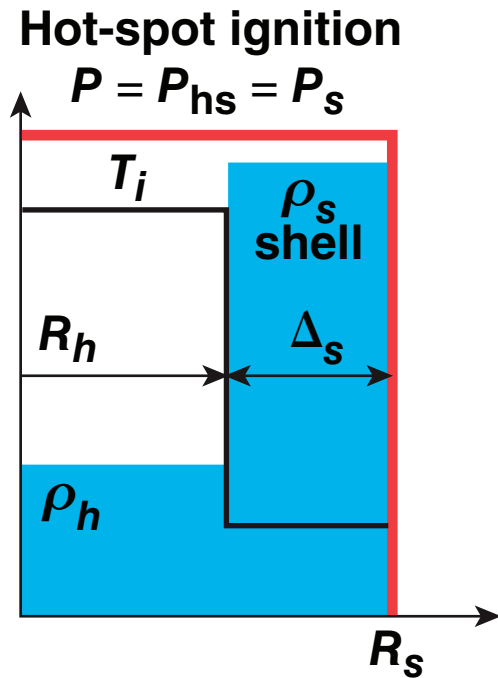


Fast-Ignition Fuel Assembly: Theory and Experiments



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Summary

Scaling laws for fast-ignition fuel assembly are derived and used to design high-density and high-areal-density implosions



- High-density and high-areal-density capsules are optimized for fast-ignition implosions.
- Density depends on adiabat and implosion velocity. It is independent of driver energy.
- Areal density depends on adiabat and driver energy, and depends weakly on implosion velocity.
- Hot-spot temperature depends only on the implosion velocity.
- Low-adiabat, low-implosion-velocity cryogenic implosions on OMEGA can achieve areal densities up to 0.78 g/cm^2 .

Energy gain increases for low-implosion velocity and high areal density



$$G = \frac{\theta E_f / m_{\text{ion}}}{V_i^2 / \eta_h} = \frac{\eta_h}{V_i^2} \frac{\theta}{E_f m_{\text{ion}}}$$

$$\theta = \frac{1}{1 + 7 / \rho R} = \text{fraction burned}$$

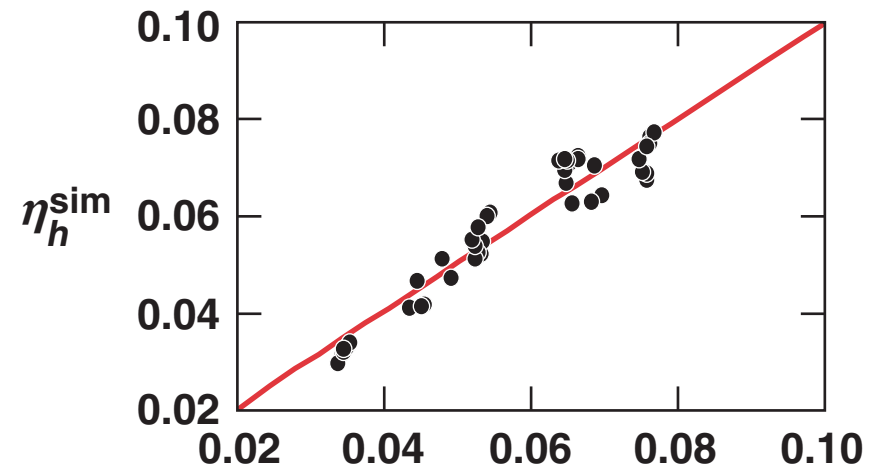
m_i = ion mass

$E_f = 17.5$ MeV

η_h = hydrodynamic efficiency

Gain formula \Rightarrow
$$G = \frac{73}{I_{15}^{0.25}} \left(\frac{3 \times 10^7}{V_i} \right)^{1.25} \left(\frac{\theta}{0.2} \right)$$

$$\eta_h^{\text{theory}} \sim V_i^{0.87} I_L^{-0.29}$$



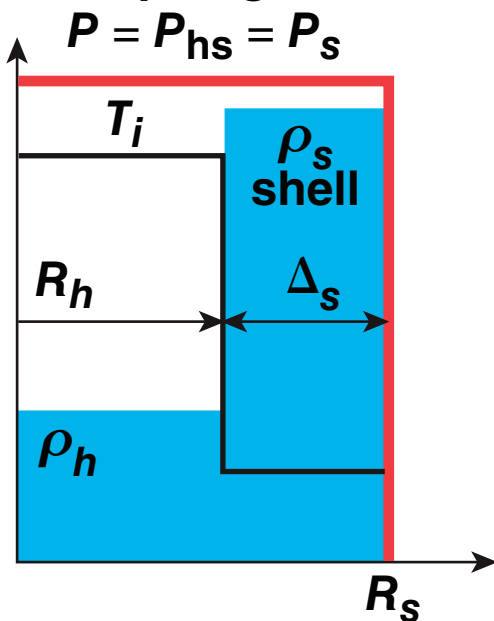
$$\eta_h^{\text{fit}} = \frac{0.049}{I_{15}^{0.25}} \left[\frac{V_i (\text{cm/s})}{3 \times 10^7} \right]^{0.75}$$

- Higher $\rho R \rightarrow$ longer confinement time
- Lower $V_i \rightarrow$ more fuel mass for the same kinetic/laser energy

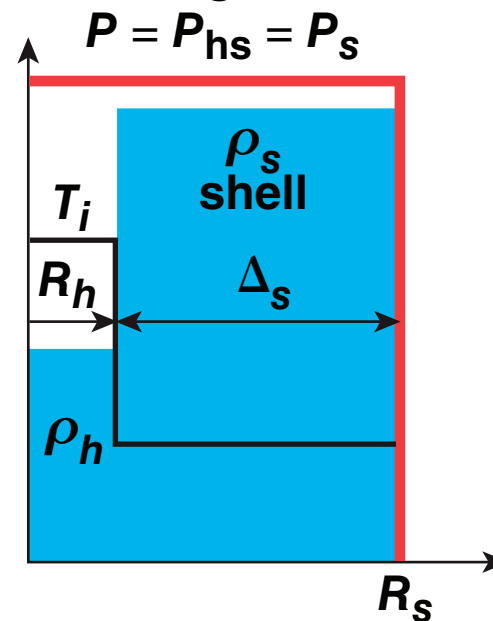
Scaling laws relating stagnation properties to in-flight hydrodynamic variables are derived from conservation equations



Hot-spot ignition



Fast ignition



Mass: $\rho_s \Delta_s \sim \frac{M_{sh}}{R_h^2 \Sigma(A_s)} \sim \frac{E_k}{R_h^2 V_i^2 \Sigma(A_s)}$

Energy: $E_k \sim P_s (R_h + \Delta_s)^3$

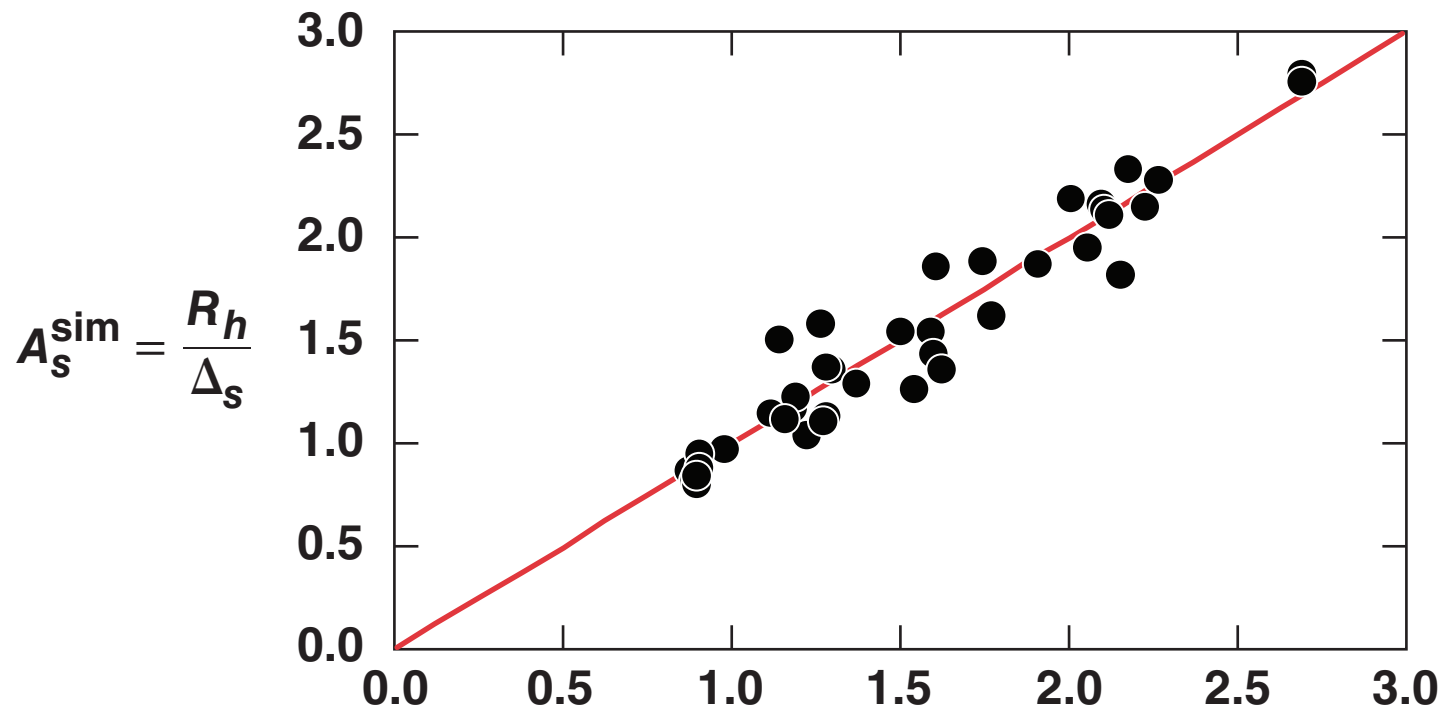
Entropy:* $\alpha_s \sim \alpha_{if} \text{Mach}_{if}^{2/3}$

Aspect ratio: $A_s = \frac{R_h}{\Delta_s}$

Volume factor: $\Sigma(x) \equiv 1 + \frac{1}{x} + \frac{1}{3x^2}$

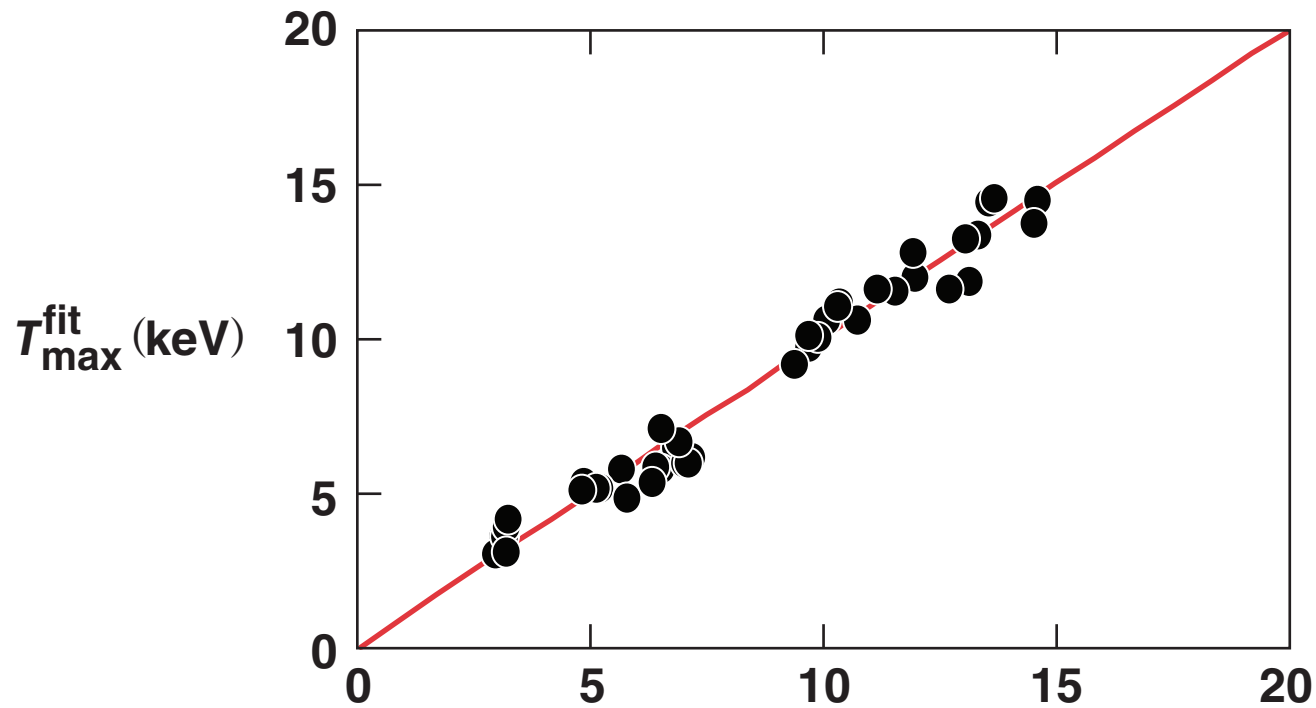
Unknowns: $\rightarrow P_s, \rho_s, A_s, \Delta_s$

The stagnation aspect ratio decreases with lower implosion velocity



$$A_s^{\text{fit}} = 2.1 \left[\frac{V_i (\text{cm/s})}{3 \times 10^7} \right]^{0.96}$$

The hot-spot temperature decreases with lower velocity

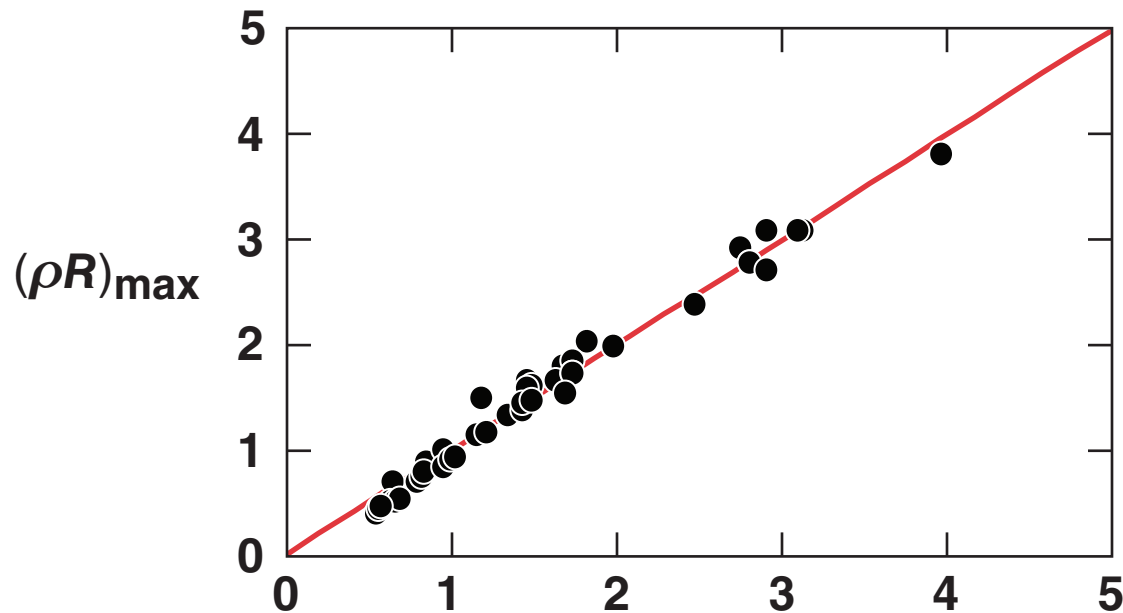


$$T_{\text{hot spot}}^{\text{max}} (\text{keV})^{\text{fit}} = 7 \left[\frac{V_i (\text{cm/s})}{3 \times 10^7} \right]^{1.4} \alpha^{-0.04}$$

The areal density is dependent on adiabat and driver energy



$$(\rho R)^{\text{theory}} \sim E_L^{0.33} \alpha_{\text{if}}^{-0.8} V_I^{0.03}$$



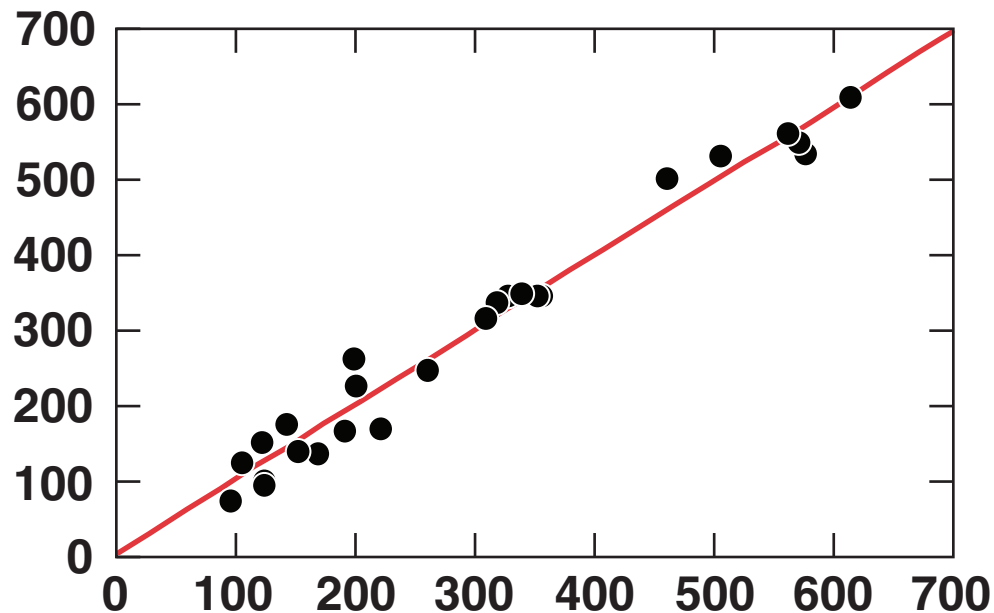
$$(\rho R)_{\text{max}}^{\text{fit}} = \frac{1.2}{\alpha^{0.57}} \left[\frac{E_L \text{ (kJ)}}{100} \right]^{0.33} \left[\frac{V_i \text{ (cm/s)}}{3 \times 10^7} \right]^{0.1}$$

Fast ignition requires large enough densities; the density depends on velocity and adiabat



$$\rho_s^{\text{theory}} \sim V_I^{1.4} \alpha_{\text{if}}^{-1.2}$$

$\langle \rho \rangle_{\rho R}$



$$\langle \rho R \rangle_{\rho R}^{\text{fit}} = \frac{440}{\alpha^{1.03}} \left[\frac{V_i (\text{cm/s})}{3 \times 10^7} \right]^{0.93}$$

The hydrodynamics of fast ignition depend on three parameters: gain, density, and areal density



$$\text{Gain} \sim V_i^{-1.25} (1 + 7/\rho R)^{-1} \Rightarrow \frac{743}{1 + 30/E_L^{1/3} \text{ (kJ)}}$$

$$\rho R \sim E_L^{0.33} / \alpha^{0.57}$$

$$\rho \sim V_i / \alpha$$

- Fast-ignition implosion
 - low-velocity V_i
 - low-adiabat α
 - large mass

$$E_{ig}^* \text{ (kJ)} \approx 11 \left[\frac{400}{\rho \text{ (g/cc)}} \right]^{1.95}$$

High ρ is required for fast ignition

$$r_{\text{beam}}^* \text{ (\mu m)} = 15 \left[\frac{400}{\rho \text{ (g/cc)}} \right]^{0.95}$$

Upper bound of the density

Low-adiabat implosions lead to high ρ and ρR with low velocities, large masses, and high gains



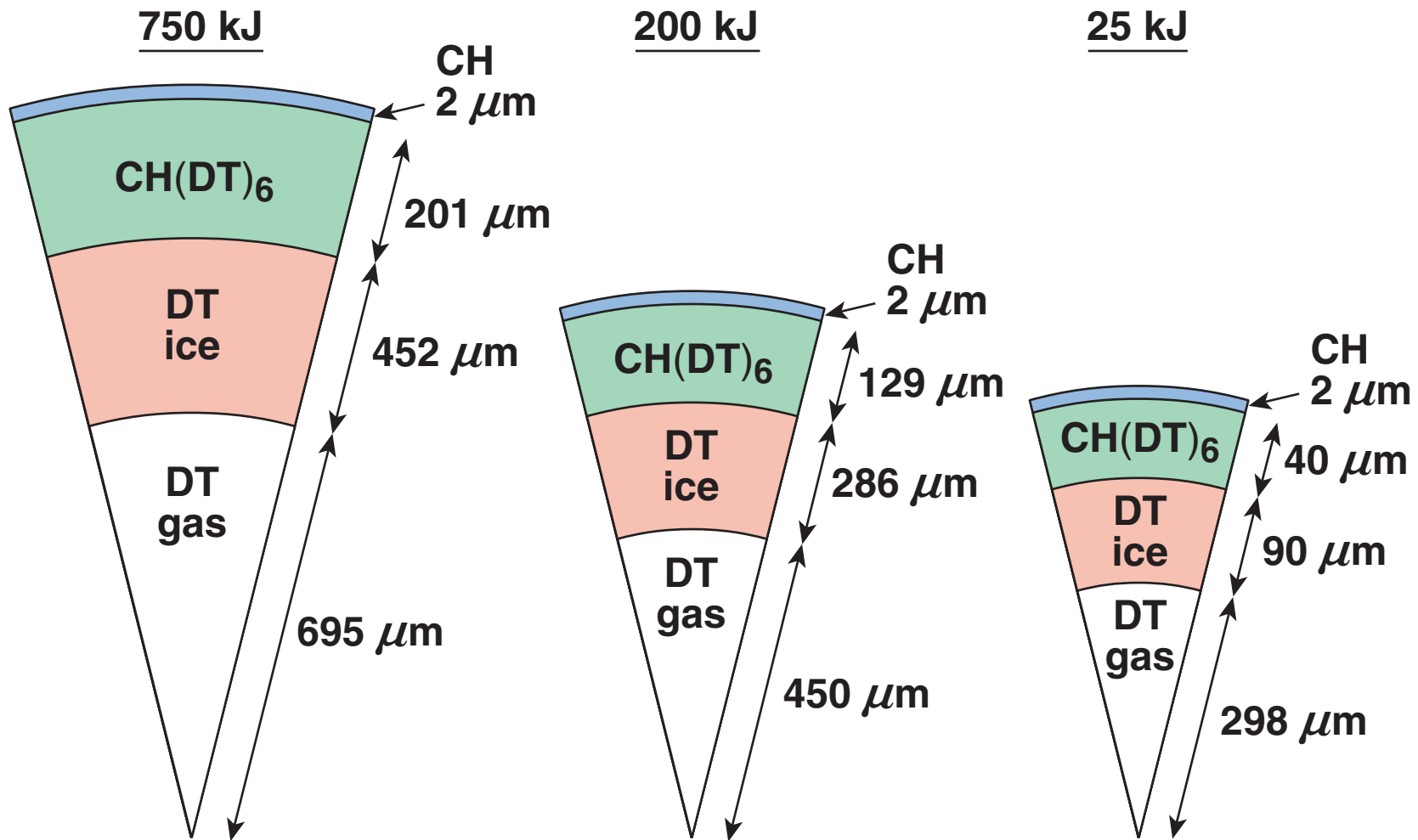
Implosion Characteristics

- Choose the lowest possible adiabat. Limitation to the minimum adiabat comes from the laser pulse length and the pulse contrast ratio; $\alpha = 0.7$ seems a reasonable value
- Choose stagnation density
- Find the implosion velocity from the density equation

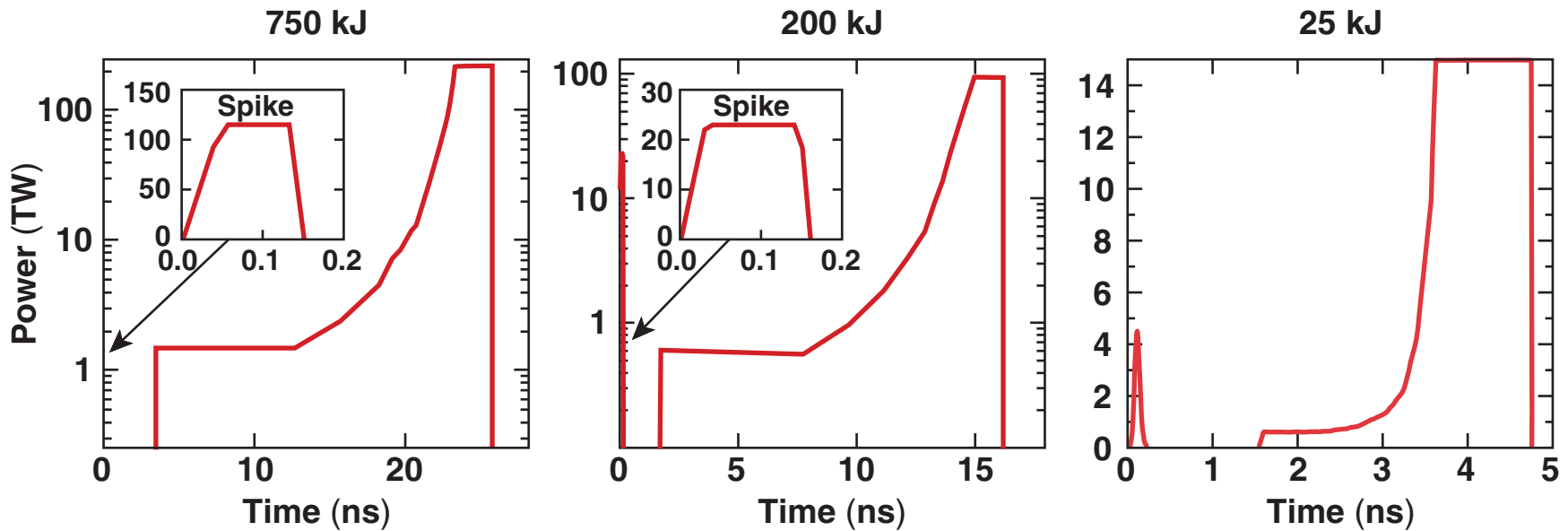
Target Design

- Set $I \approx 10^{15}$ W/cm²
- Choose driver energy and corresponding laser power
- Find capsule outer radius from power and intensity
- Find final mass from kinetic energy
- Assuming a 20% ablated mass leads to an initial mass
- Initial mass and outer radius yield the inner radius

Optimized fast-ignition cryo targets are thick shells of wetted foam with an initial aspect ratio of ~ 2



These targets have high areal densities and low IFAR



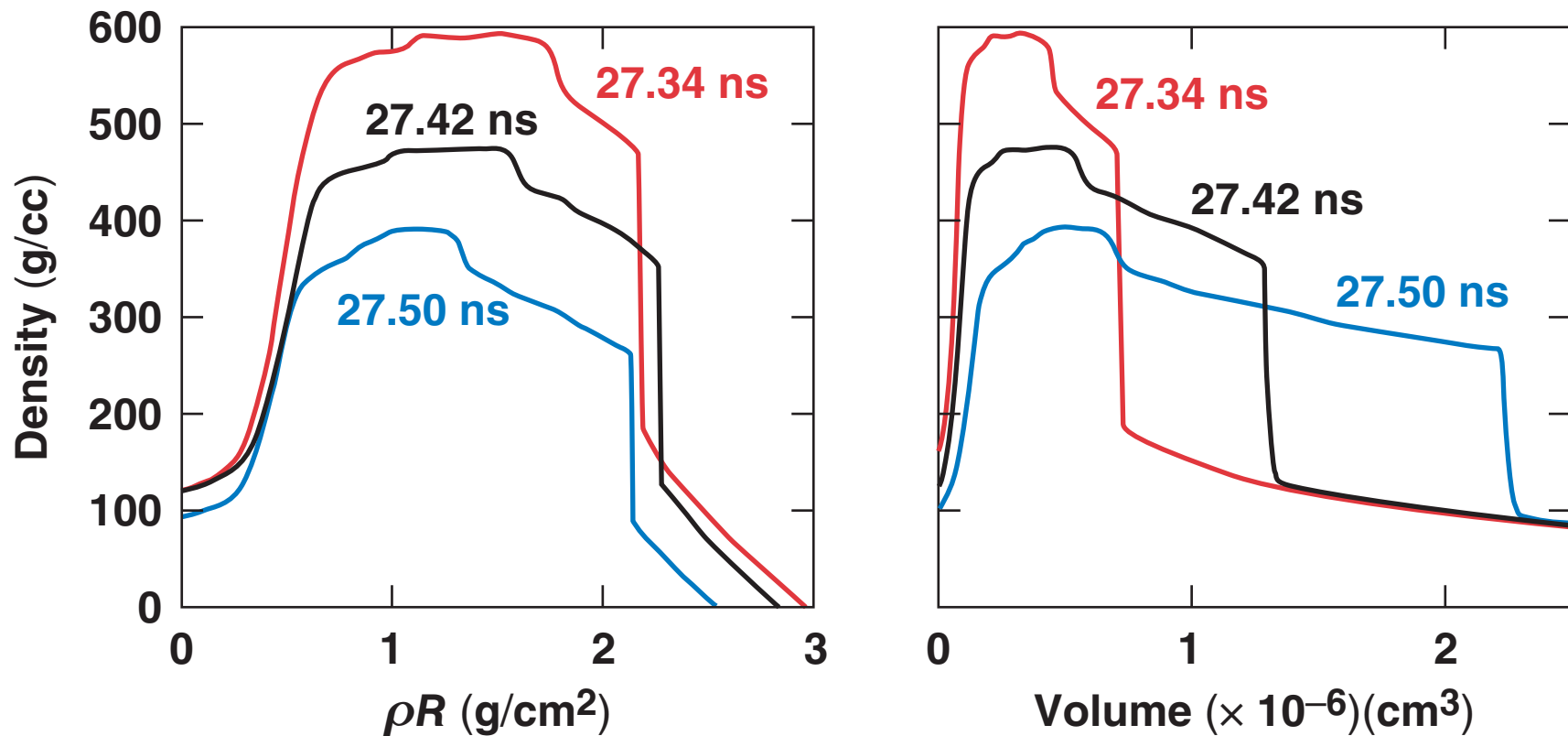
Maximum ρR	3 g/cm ²
α	0.7
V_j	1.7×10^7 cm/s
IFAR	18

Maximum ρR	1.9 g/cm ²
α	0.7
V_j	1.7×10^7 cm/s
IFAR	18

Maximum ρR	0.78 g/cm ²
α	1.0
V_j	2.6×10^7 cm/s
IFAR	30

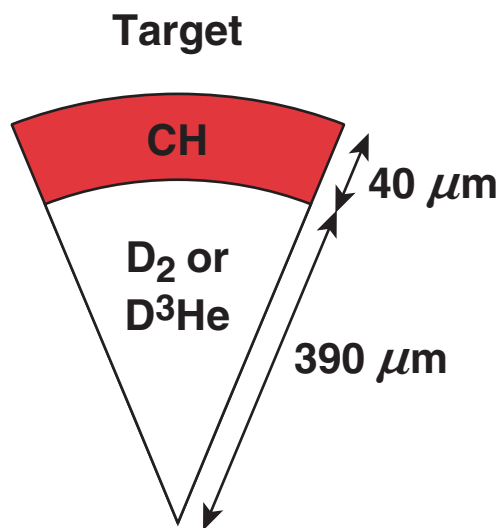
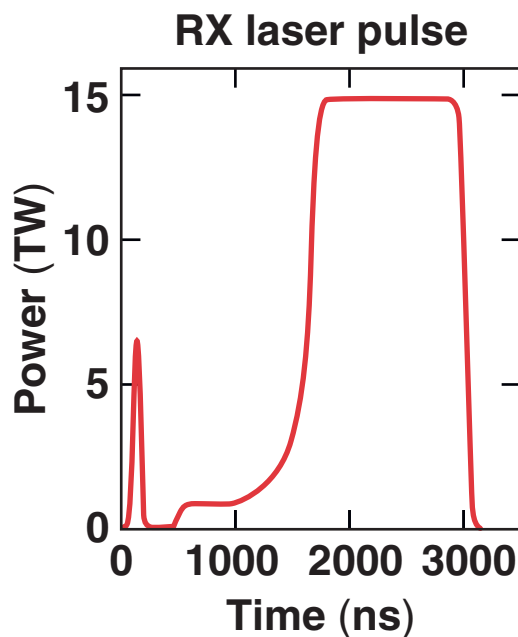
Low-adiabat implosions are driven by RX laser pulses.

The 750-kJ capsule yields a density >300 g/cc over a $\rho R > 2$ g/cm²



The hot-spot volume is $<8\%$ of the compressed volume.

Low-adiabat, low- V_i implosions of surrogate CH targets are used to study fast-ignition fuel assemblies on OMEGA

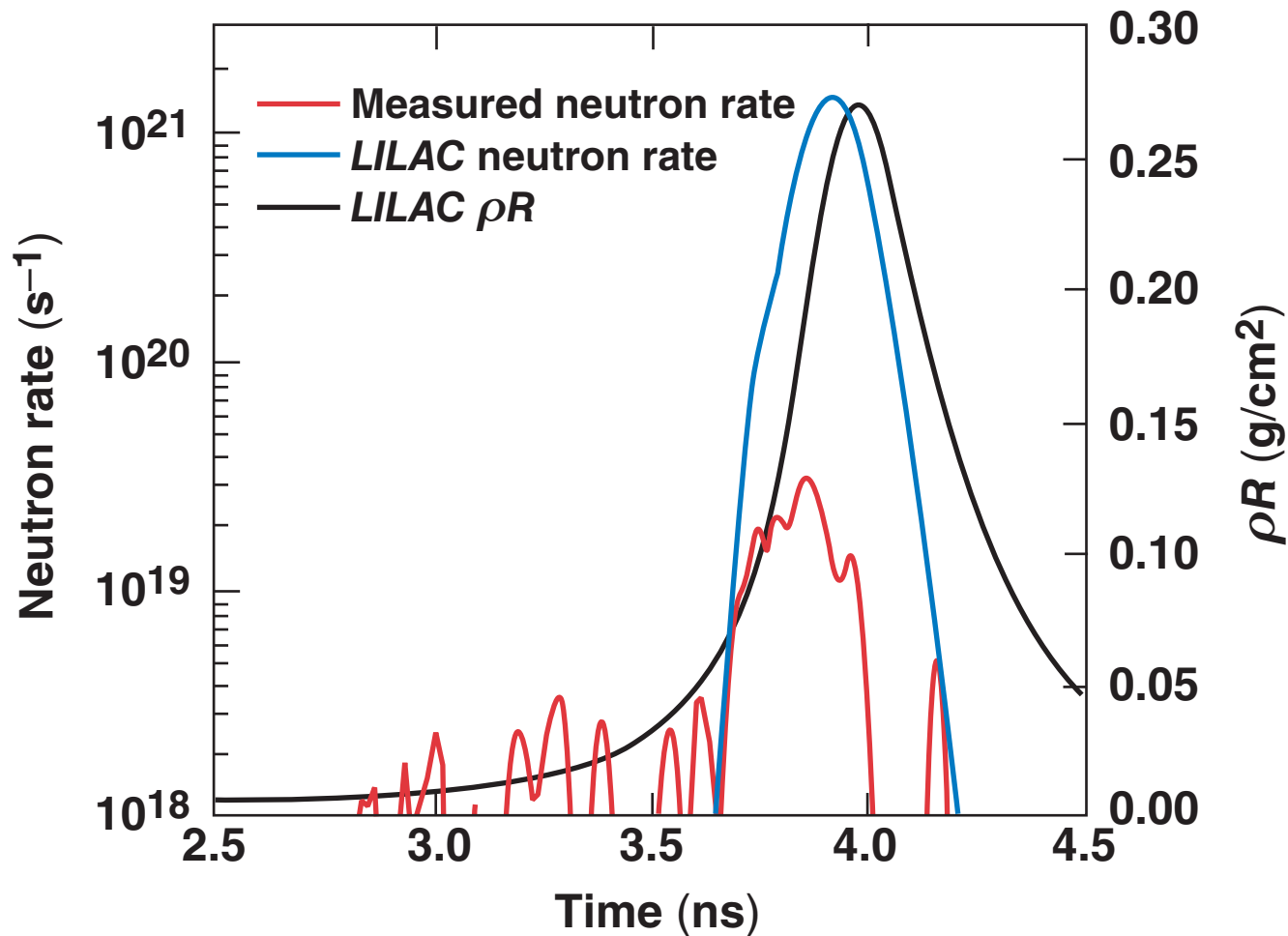


Simulated values

Pressure (atm)	ρ (g/cc)	$2\rho R$ (g/cm ²)	E_{stop} (MeV)*
35	89.3	0.52	2.10
25	101	0.62	2.35
15	120	0.74	2.76
5	185	1.02	3.53
1	222	1.32	4.45

$E_L \approx 20 \text{ kJ}, \alpha \approx 1.3,$
 $V_i \approx 2 \cdot 10^7 \text{ cm/s}$

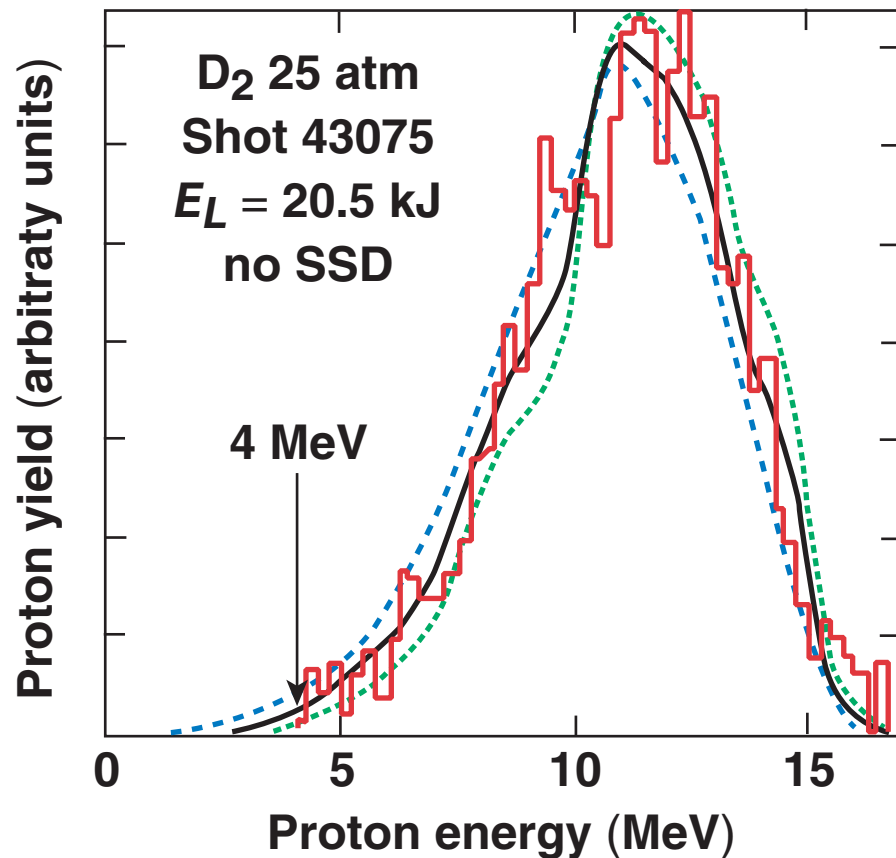
The DD neutron production begins as predicted and shows a 200-ps truncation, probably due to hot-spot CH–DD mixing



D₂ 25 atm
Shot # 43075
 $E_L = 20.5$ kJ
No SSD

YOC = 3%

The measured¹ and reconstructed² downshifted secondary proton spectra are in good agreement



The reconstruction used the measured neutron rate and the simulated $\rho R(t)$.

- Measured spectrum
- - - LILAC with 1-D source size
- - - LILAC with point source
- LILAC averaged

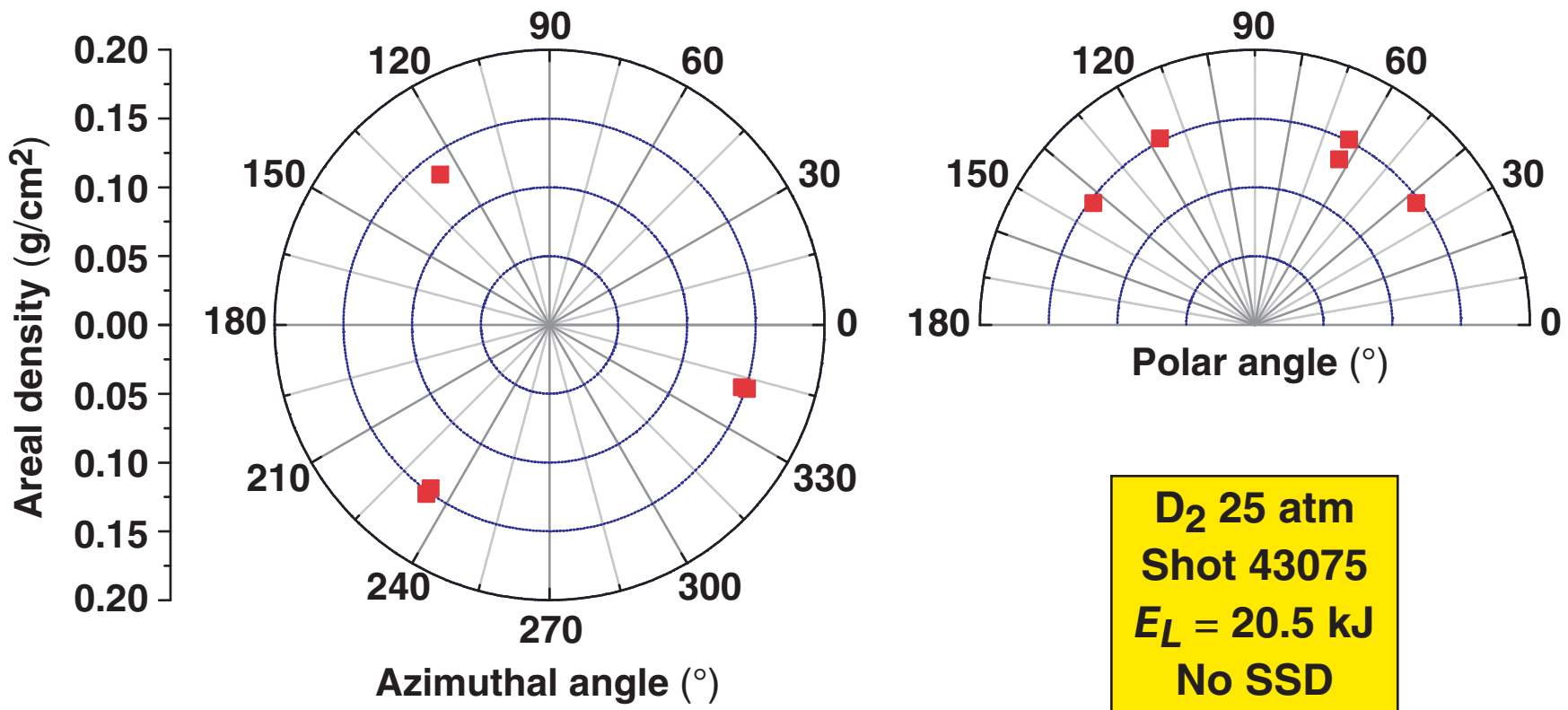
8.7-MeV downshift



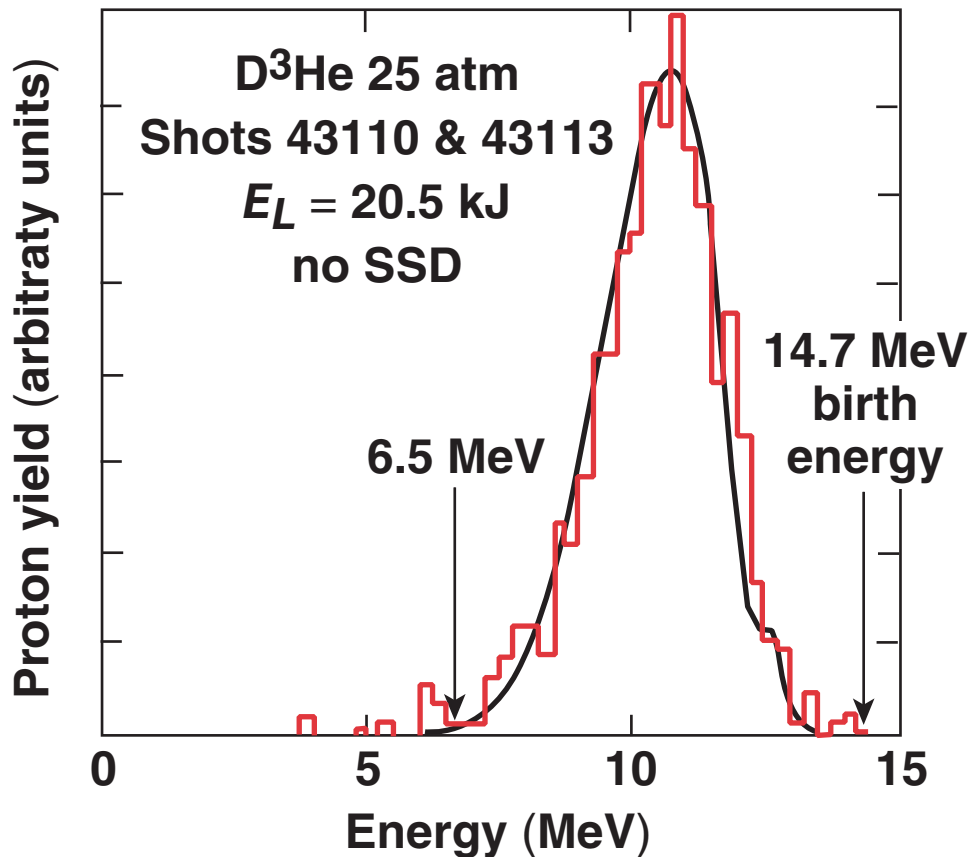
$$(\rho R)_{\max} = 0.26 \text{ g/cm}^2$$
$$\langle \rho R \rangle_n = 0.15 \text{ g/cm}^2$$

¹F. H. Séguin *et al.*, Rev. Sci. Instrum. **74**, 975 (2003).
²P. B. Radha *et al.*, APS/DPP 2006 (GO2.00008).

The $\langle \rho R \rangle$ modulations are $<10\%$, indicating that the compressed core is not significantly affected by low-mode ($\ell \leq 5$) nonuniformities



The measured¹ and reconstructed² downshifted primary proton spectra are in good agreement for D³He implosions



The calculation used an estimated proton rate and the simulated ρR evolution

— Measured spectrum
— Calculated spectrum

8.2-MeV downshift

↓

$(\rho R)_{\max} = 0.24$ g/cm²
 $\langle \rho R \rangle_n = 0.13$ g/cm²

¹F. H. Séguin *et al.*, Rev. Sci. Instrum. **74**, 975 (2003).
²P. B. Radha *et al.*, APS/DPP 2006 (GO2.00008).

Very good agreement between measured and predicted areal densities is obtained



Shot number	Gas fill	Pressure (atm)	Measured $\langle \rho R \rangle_p$	Sim. shell $\langle \rho R \rangle_n$	Measured ρR_n^{\max}	Sim. ρR^{\max}
43074	D ₂	34	0.133	0.138	0.25	0.24
43075	D ₂	25	0.146	0.144	0.26	0.26
43107	D ₂	25	0.122	0.132	0.24	0.27
43114	D ₂	25	0.128	0.112	0.23	0.23
43106	D ₂	13	0.128	—	—	—
43108	D ₂	13	0.129	—	—	—
43109 + 43112	D ³ He	33	0.128	—	0.24	0.25
43110 + 43113	D ³ He	25	0.130	—	0.24	0.28
Average			0.131	0.132	0.24	0.25